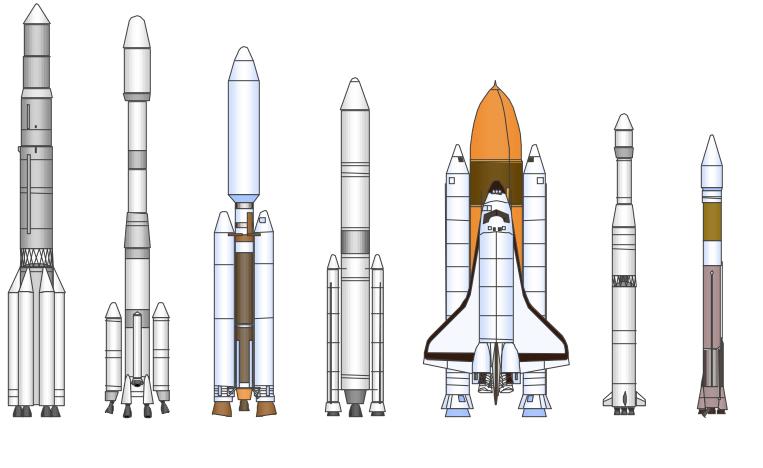
30-Day Launch Forecast 3 August 2000



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Distribution

XOO: AF/CC AF/CV AF/INXY AF/TEP AF/XO AF/XOI AF/XORR AF/XOOOO (AFDO) AF/XOOOOB AF/XPPS BMDO/TRT DAMO-FDW J-38/DSOD J-38/NOD-NOB J-5/POL J-6S NAIC/IA NAIC/IAS OSD/C3I OSD/DDR&E SAF SAF/OSX SAF/SX SAF/SXP 497th IG/INOA

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30-Day Launch Forecast (03 August 2000 - 01 September 2000)

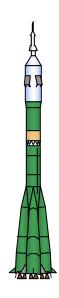
Mon	Tue	Wed	Thu	Fri	Sat	Sun	Comments / Schedule Changes
A Look Ahead 08 Sep 14 Sep 18 Sep 21 Sep 24 Sep 28 Sep 28 Sep All foreign launche	STS-106 Titan 2 Sea Launch Soyuz-U Zenit 2 Minuteman III Minuteman III	ISS 2A.2b NOAA-L Thuraya-1A Progress M1 Badr-2 GT-173-GM GT-174-GM recast are unofficial	3	4	5	Soyuz-U Progress M1 Baikonur 1422 EDT	Soyuz-U / Progress M1-3 • Resupply for ISS Soyuz-Fregat / Cluster II • Second pair of four identically instrumented science satellites sponsored by ESA
7	8	Soyuz- Fregat Cluster II Baikonur 0711 EDT	10	11	12	13	Titan 4B / NRO; Mission B-28 • Classified military satellite • No Upper Stage (NUS/403 configuration)
14	15	16 Titan 4B NRO SLC-4E VAFB 1800-2200 EDT	Ariane 44LP Brasilsat B-4 Nilesat 102 ELA-2 CSG TBD EDT	18	19	20	Ariane 44LP / Brasilsat B-4 / Nilesat 102; Flight 131 • Brasilsat B-4: Brazilian communications satellite • Nilesat 102: Egyptian communications satellite Delta 3 / DM-F3; Flight 280 • Demonstration flight with dummy payload to prove vehicle flightworthiness
21	22	23	24	Dnepr Saudisat 1-A & 1-B Baikonur TBD EDT	26	27	Dnepr / Saudisat 1-A &1-B / UNISAT / Megsat 1 / TiungSat 1 • Saudisat: Saudi Institute for Space Research • UNISAT: University of Rome microsatellite • Megsat 1: Italian data relay satellite • TiungSat 1: Malaysian remote sensing payload
28	29	30	31	1 Sep	Last Week's Lat Date Vehicle 28 Jul Sea Launce Launch Date provided	<u>Payload</u> ch PAS-9	<u>Site</u> <u>Type</u> Pacific Ocean Communications

Acronyms: VAFB - Vander

VAFB - Vandenberg AFB CA SLC - Space Launch Complex

CCAFS - Cape Canaveral AFS FL LC- Launch Complex KSC - Kennedy Space Center FL LF - Launch Facility CSLF - Calif. Space Launch Facility EDT - Eastern Daylight Time NET - No Earlier Than EST - Eastern Standard Time WFF - Wallops Flight Facility CSG - Guiana Space Center

Soyuz-U



Current Mission Specifics

551st launch of a Soyuz Launch Vehicle since 1980

Reliability History (since 1980)

• 537 successes in 550 attempts

Typical Launch Sequence

Lift off	0 se	ec.
 Strap-ons separate 	118 se	ec
 Payload fairing jettison 	160 se	ec
 Core stage 1 separation 	286 se	ec
Orbit Injection	540 se	ec

Payload Weight: Progress M1; 16,095 lb (at launch)

Orbit: 210 nm circular, 51.6° inclination

Next Soyuz launch

• 9 August 2000 / Cluster II

Background Information

First Launch: November 1963

Flight Rate: 45 per year (maximum recorded launch rate)
Launch Site: Plesetsk, Russia; Baikonur, Kazakhstan

Capability: 15,400 lb to LEO

History

- Developed from the Vostok Launch Vehicle originally derived from the SS-6 (Sapwood) ICBM.
- Used to launch every former Soviet Union piloted spacecraft since 1964.
- Also used to launch photo reconnaissance satellites, earth resource satellites, and Progress resupply missions to the Mir space station.

Description

- Two-stage (plus 4 strap-ons) liquid fueled vehicle.
- Stage 1 core has one RD-108 booster engine (one turbopump with four separate combustion chambers) burning LOX/kerosene fed from stage 1 tanks, generating 220.050 lb of thrust.
- Four Stage 1 strap-ons each have one RD-107 engine (one turbopump with four separate combustion chambers) burning LOX/kerosene fed from stage 1 tank, generating a total of 227,925 lb of thrust each.
- Stage 2 has one RD-0110 Block 1 engine burning LOX/kerosene, generating 67,050 lb of thrust.
- Starsem version only: Fregat restartable upper stage powered by a single-chamber Lavochkin engine engine burning UDMH/N₂O₄, generating 4,410 lb of vacuum thrust.

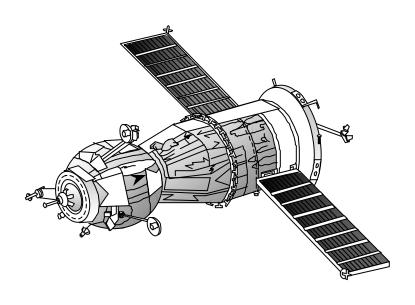
Profile

 Length:
 162.5 ft
 Launch Weight:
 682,765 lb

 Diameter:
 33.8 ft
 Liftoff Thrust:
 1,334,700 lb

Payload Fairing: 37.3 ft x 9.8 ft

Progress-M1



Spacecraft Specifications

Total Weight:

• 16,095 lb (at launch)

Dimensions:

Total Length: 23.7 ft
 Diameter: 7.2 ft
 Solar Arrays: 107.6 ft²

Progress-M1

Unmanned Russian space station resupply ferry.

Mission

Deliver removable cargo such as experimental equipment, food, water, and air regeneration cylinders as well as propellants and compressed gases to the International Space Station (ISS).

Description

Spacecraft Description:

- Crew-accessible cargo compartment; mid-section compartment for fuel components; aft section instrument assembly module.
- Carries 3,970 lb of consumables in cargo compartment and 2,072 lb of fuel in mid-section (UDMH/NTO and O₂).
- TM-type solar arrays generating 1.3 kW; batteries carried in pressurized instrument package augment Mir when docked.
- Docks with ISS employing Kurs ('course') approach system; station crew docking assist available.
- Navigation computations performed by mission control and uplinked for execution.
- Raduga coni-cylindrical re-entry capsule used to return film and experimental samples; separates during re-entry for parachute recovery.
- Design life: 1 time use only (destructive re-entry over Pacific Ocean).

Orbit: 210 nm circular, 51.6° inclination

Prime Contractor: RKK Energia

Cluster II



Spacecraft Specifications

Weight:

- 2,645 lb (at launch)
- 1,213 lb (dry mass)

Dimensions:

Height: 4.3 ft Diameter: 9.5 ft

Cluster II

Second pair of four identical satellites that will fly in formation. Cluster II is one of ESA's top priority Cornerstone science missions, and replaces the original Cluster mission that was destroyed during the failed maiden launch of the Ariane 5 rocket in June 1996.

Mission

Study the interaction between the the solar wind and the Earth's magnetosphere allowing for the first time truly three-dimensional measurements of both large- and small-scale phenomena in the near-Earth environment.

Description

Spacecraft Description:

- Spin-stabilized cylindrical bus; orbit/attitude maintenance performed by semi-radial and axial control thrusters together with the main engine.
- Each satellite carries an identical set of 11 instruments mounted to the Main Equipment Platform.
- Power: 224 W provided by six curved solar-array panels; five 80 Ah Silver Cadmium batteries provide eclipse protection.
- Carries two 5 meter-long experiment booms, four 50 meter-long wire booms, and two antenna booms.
- Telemetry downlink bit rate 2 to 262 kbit/s.
- Design life: 2 years.

Orbit: Highly eccentric polar orbits ranging from 13,510 to 67,555 nm at 64.8° - 90° inclination

Prime Contractor: Dornier

Titan 4B



Current Mission Specifics

30th launch of the Titan 4 launch vehicle

Reliability History:

- Titan 4-A: 20 successes in 22 attempts
- Titan 4-B: 5 successes in 7 attempts

Typical Launch Sequence

Stage 0 Ignition	0	sec
Stage 1 Ignition	120	sec
Stage 0 Separation	152	sec
 Payload Fairing Separation 	230	sec
Stage 2 Ignition	308	sec
Stage 1 Separation	309	sec
Stage 2 Shutdown	532	sec
Stage 2 Jettison	558	sec

Payload Weight: Classified NRO (B-28); N/A

Next Titan 4B launch

• 31 October 2000 / Milstar-4 (B-41)

Background Information

First Launch: February 1997
Flight Rate: 3-4 per year

Launch Sites: SLC-40 & SLC-41 (CCAFS, USA); SLC-4E (VAFB, USA)

Capability: 48,280 lb to LEO; 39,000 lb to polar LEO;

19,000 lb to GTO; 12,725 lb to GEO (with Centaur)

History

- USAF Titan ICBM program started in 1950s.
- Titan 4 program started as Complimentary Expendable Launch Vehicle (CELV).
- · Direct descendant of Titan 34D.
- Titan 4B program adds Solid Rocket Motor Upgrade. SRMU changes from SRM: increases payload mass capability by 25%, graphite composite casing (vs steel), HTPB (vs PBAN), 3 field joints (vs 7), 12% heavier, 5% wider.

Description

- Two or three-stage (including Centaur or IUS) vehicle plus two strap-ons.
- Stage 0: Two 3-segment Alliant Techsystems SRMUs, generate 1,701,450 lb of thrust each
- Stage 1: Two Aerojet LR-87 engines burn N₂O₄/A-50, generating 547,605 lb of thrust total
- Stage 2: One Aerojet LR-91 engine burns N₂O₄/A-50, generating 106,200 lb of thrust.
- Centaur upper stage uses two Pratt & Whitney RL10-3-3A engines burning LOX/LH₂ producing 33,000 lb of total thrust.
- Titan 4 available in three configurations: Titan 401/Centaur Upper Stage; Titan 402/Inertial Upper Stage (IUS); Titan 403/No Upper Stage (NUS).

Profile

Length: 204 ft Launch Weight: 2,070,785 lb (Centaur)

Diameter: 10 ft Liftoff Thrust: 3,402,900 lb Payload Fairing: 56 (NUS), 66 (IUS), 76 (IUS), and 86 (Centaur) ft x 16 ft

Space Launch Activities 2000 Year To Date

Unite	d States La	aunches			Frenc	h Launches			
Date	Vehicle	Payload	<u>Site</u>	<u>Type</u>	Date	Vehicle	<u>Payload</u>	<u>Site</u>	<u>Type</u>
18 Jan	Minuteman II	IFT-4	VAFB, LF-03	Missile Defense (MIL)	25 Jan	Ariane 42L	Galaxy-10R	CSG, ELA-2	Communications (COM)
21 Jan	Atlas 2A	DSCS-B8	CCAFS, SLC-36A	Communications (MIL)	18 Feb	Ariane 44LP	SUPERBIRD-4	CSG, ELA-2	Communications (COM)
27 Jan	Minotaur	JAWSAT	VAFB, SLC-7	Technology Demo (MIL)	21 Mar	Ariane 505	INSAT-3B/	CSG, ELA-3	Communications (COM)
03 Feb	Atlas 2AS	Hispasat 1-C	CCAFS, SLC-36B	Communications (COM)			AsiaStar		Communications (COM)
08 Feb	Delta 2	Globalstar-14	CCAFS, SLC-17B	Communications (COM)	19 Apr	Ariane 42L	Galaxy 4-R	CSG, ELA-2	Communications (COM)
11 Feb	STS-99	SRTM	KSC, LC-39A	Radar Mapping (CIV)					
08 Mar	Peacekeeper	GT-29-PA	VAFB, LF-05	FOT&E (MIL)					
12 Mar	Taurus	MTI	VAFB, 576-E	Technology Demo (MIL)					
12 Mar*	Sea Launch	ICO F-1	Pacific Ocean	Communications (COM)					
25 Mar	Delta 2	IMAGE GOES-L	VAFB, SLC-2W	Science (CIV)					
03 May 08 May	Atlas 2A Titan 4B	DSP-20	CCAFS, SLC-36A CCAFS, SLC-40	Meteorology (CIV) Early Warning (MIL)					
11 May	Delta 2	GPS IIR-4	CCAFS, SLC-40	Navigation (MIL)	Chine	se Launche	9		
19 May	STS-101	ISS 2A.2a	KSC, LC-39A	ISS Resupply (CIV)			_	C:4-	Toma
24 May	Minuteman III		VAFB. LF-09	Flight Test Missile (MIL)	<u>Date</u>	<u>Vehicle</u>	Payload	<u>Site</u>	Type
24 May	Atlas 3A	Eutelsat-W4	CCAFS, SLC-36B	Communications (COM)	25 Jan 25 Jun	LM 3A LM 3	Zhongxing-22	Xichang	Communications (CIV)
07 Jun	Pegasus XL	TSX-5	VAFB	Science (MIL)	25 Jun	LIVI 3	Fengyun-2B	Xichang	Meteorological (CIV)
09 Jun	Minuteman III	GT-172-GM	VAFB, LF-10	FOT&E (MIL)					
30 Jun	Atlas 2A	TDRS-H	CCAFS, SLC-36A	Communications (CIV)					
08 Jul	Minuteman II	IFT-5	VAFB, LF-03	Missile Defense (MIL)					
14 Jul	Atlas 2AS	EchoStar-6	CCAFS, SLC-36B	Communications (COM)					
16 Jul	Delta 2	GPS IIR-5	CCAFS, SLC-17A	Navigation (MIL)	Indian	Launches			
19 Jul	Minotaur	MightySat II.1	VAFB, CSLF	Technology Demo (MIL)	Date	<u>Vehicle</u>	Payload	<u>Site</u>	Type
28 Jul	Sea Launch	PAS-9	Pacific Ocean	Communications (COM)	No Laune	hes to Date	-		
					NO Lauric	nes to Date			
					Japan	ese Launch	es		
					Date	Vehicle	Payload	<u>Site</u>	<u>Type</u>
					10 Feb*	M-5	ASTRO-E	Kagoshima	Science (CIV)
					10 1 60	WFS	ASTRO-L	Nagosiiiila	ocience (CIV)
					Brazili	ian Launche	es		
					<u>Date</u>	<u>Vehicle</u>	<u>Payload</u>	<u>Site</u>	<u>Type</u>
					No Laund	hes to Date			

^{*} Indicates Launch Failure Launch Date provided in Universal Time

Space Launch Activities

2000 Year To Date

Russ	ian Launch	es		
<u>Date</u>	<u>Vehicle</u>	<u>Payload</u>	<u>Site</u>	<u>Type</u>
01 Feb	Soyuz-U	Progress M1-1	Baikonur	Mir Resupply (CIV)
03 Feb	Zenit 2	Cosmos 2369	Baikonur	Signal Intelligence (MIL)
08 Feb	Soyuz-Fregat	IRDT	Baikonur	Technology Demo (COM)
12 Feb	Proton	Garuda-1	Baikonur	Communications (COM)
12 Mar	Proton	Express-6A	Baikonur	Communications (CIV)
20 Mar	Soyuz-Fregat	Dumsat	Baikonur	Technology Demo (COM)
04 Apr	Soyuz-U	Soyuz TM-30	Baikonur	Mir Resupply (CIV)
17 Apr	Proton	SESat	Baikonur	Communications (COM)
25 Apr	Soyuz-U	Progress M1-2	Baikonur	Mir Resupply (CIV)
03 May	Soyuz-U	Cosmos 2370	Baikonur	Classified (MIL)
16 May	Eurockot	SIMSAT-1 & -2	Plesetsk	Demo Flight (COM)
06 Jun	Proton	Gorizont-45	Baikonur	Communications (CIV)
24 Jun	Proton	Express-3A	Baikonur	Communications (CIV)
28 Jun	Kosmos 3M	Nadezhda/	Plesetsk	Navigation (CIV)
		Tsinghua-1/		Remote Sensing (CIV)
		SNAP-1		Technology Demo (CIV)
30 Jun	Proton	Sirius-1	Baikonur	Communications (COM)
04 Jul	Proton	Cosmos 2371	Baikonur	Data Relay (MIL)
12 Jul	Proton	Zvezda	Baikonur	ISS (CIV)
15 Jul	Kosmos 3M	CHAMP/	Plesetsk	Science (CIV)
		MITA/		Technology Demo (CIV)
		RUBIN		Science (CIV)
16 Jul	Soyuz-Fregat	Cluster II	Baikonur	Science (CIV)

Launch Market Analysis

By Country

	# of Launches	Percent of Market
US	13	35.1%
Russia	17	45.9%
France	4	10.8%
China	2	5.4%
Japan	1	2.7%

By Mission

	# of Launches	Percent of Market
US Military	4	10.8%
US Civil	3	8.1%
US Commercial	6	16.2%
World Military	3	8.1%
World Civil	11	29.7%
World Commercial	10	27.0%

By Orbit Type (Commercial Only)

<u>GEO</u>	# of Launches	Percent of Market
US	4	40.0%
Russia	2	20.0%
France	4	40.0%
China	0	0.0%
Japan	0	0.0%
<u>LEO</u>	# of Launches	Percent of Market
LEO US	# of Launches 2	Percent of Market 33.3%
US	2	33.3%
US Russia	2 4	33.3% 66.7%

Figures Do Not Include US Space Shuttle, Small Launch Vehicles, or ICBM launches